

# What is Bernoulli's Principle?

## Worksheet

Bernoulli's principle states that for an ideal, incompressible, non-viscous fluid flowing steadily, the sum of static pressure, dynamic pressure, and hydrostatic pressure is constant along a streamline - so as speed increases, pressure decreases.

$$P + \frac{1}{2}\rho v^2 + \rho g h = \text{constant}$$

## Questions

- As fluid speed increases in a horizontal pipe, what happens to pressure?
  - It increases
  - It decreases
  - It stays the same
  - It becomes zero
- Bernoulli's equation assumes the fluid is...
  - Compressible and viscous
  - Incompressible and non-viscous
  - Always at rest
  - Always turbulent
- In  $P + v + gh = \text{constant}$ , the term  $gh$  represents...
  - Dynamic pressure
  - Static pressure
  - Hydrostatic (potential) pressure
  - Viscous pressure
- Why does an airplane wing generate lift?
  - Higher pressure below than above due to faster flow on top
  - Equal pressure above and below
  - Lower air density above the wing only
  - Gravity pulls air upward
- Water flows through a horizontal pipe that narrows from a wide section ( $v_1 = 2 \text{ m/s}$ ,  $P_1 = 200,000 \text{ Pa}$ ) to a narrow section ( $v_2 = 6 \text{ m/s}$ ). Find  $P_2$  ( $\rho = 1000 \text{ kg/m}^3$ ).
- Air flows over an airplane wing at  $250 \text{ m/s}$  on top and  $200 \text{ m/s}$  below. If  $P_{\text{below}} = 101,000 \text{ Pa}$  and  $\rho_{\text{air}} = 1.2 \text{ kg/m}^3$ , find the pressure above.
- A venturi meter has a wide section ( $v_1 = 1 \text{ m/s}$ ) and throat ( $v_2 = 4 \text{ m/s}$ ), both at the same height, water flowing ( $\rho = 1000 \text{ kg/m}^3$ ). Find the pressure drop  $P_1 - P_2$ .
- Define: What does Bernoulli's principle state?
- Define: What conditions does Bernoulli's equation assume?
- Define: How does Bernoulli's principle explain lift?

## Answer Key

1. B) It decreases - By Bernoulli's principle,  $P + v$  is constant, so higher  $v$  means lower  $P$ .
2. B) Incompressible and non-viscous - The ideal Bernoulli equation assumes steady, incompressible, non-viscous flow.
3. C) Hydrostatic (potential) pressure -  $gh$  is the pressure due to elevation - the hydrostatic term.
4. A) Higher pressure below than above due to faster flow on top - Faster flow over the curved top lowers pressure there, so higher pressure below pushes the wing up.
5.  $P_1 + v_1^2 = P_2 + v_2^2$   
 $200,000 + 0.510002 = P_2 + 0.510006$   
 $200,000 + 2,000 = P_2 + 18,000$   
 $P_2 = 202,000 - 18,000 = 184,000 \text{ Pa}$
6.  $P_{\text{above}} = P_{\text{below}} + \frac{1}{2}(\rho v_{\text{below}}^2 - \rho v_{\text{above}}^2)$   
 $P_{\text{above}} = 101,000 + 0.51 \cdot 2(200^2 - 250^2)$   
 $P_{\text{above}} = 101,000 + 0.6(40,000 - 62,500)$   
 $P_{\text{above}} = 101,000 - 0.6(22,500) = 101,000 - 13,500 = 87,500 \text{ Pa}$
7.  $P_1 - \frac{1}{2}\rho v_1^2 = P_2 - \frac{1}{2}\rho v_2^2$   
 $P_1 - P_2 = \frac{1}{2}\rho(v_1^2 - v_2^2)$   
 $P_1 - P_2 = 0.51000(4^2 - 1^2)$   
 $P_1 - P_2 = 500(16 - 1) = 500 \cdot 15 = 7,500 \text{ Pa}$
8. Along a streamline,  $P + \frac{1}{2}\rho v^2 + \rho gh$  stays constant - faster flow means lower pressure.
9. Steady, incompressible, non-viscous (ideal) flow along a streamline.
10. Faster airflow over a curved wing top lowers pressure there, creating a net upward force.

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